

## Introduction

This poster presents a hybrid reduced-order model capable of reproducing the behavior of the characteristics response of the system (lift and drag forces, amplitude/ frequency of the displacement) of a flow passing rotational and translation airfoil in turbulence regime in Finite Volume Method.

## Methodology: Hybrid reduced-order model

In this poster, the **separability** of a given quantity of interest  $s$  is assumed using the Proper Orthogonal Decomposition (POD):

$$s(\mathbf{x}, t) \approx \sum_{i=1}^n a_i(t) \phi_i(\mathbf{x})$$

The spatial mode  $\phi_i$  is given by  $\phi_i = \lambda_i^{-1/2} \mathbf{S} \psi_i$ .  $\mathbf{S} = [s(\mathbf{x}, t_1), \dots, s(\mathbf{x}, t_N)]$  is the snapshot matrix with  $N \geq n$ .  $\psi_i$  and  $\lambda_i$  are respectively the eigenvectors and eigenvalues of the matrix  $\mathbf{S}^T \mathbf{S}$  i.e  $\mathbf{S}^T \mathbf{S} \psi_i = \lambda_i \psi_i$ . The temporal modes  $a_j(t)$  is given by  $a_j(t) = (s, \phi_j)_{L^2}$ . The POD and the Galerkin projection are used to construct the reduced systems of both the velocity and pressure. In order, to build a complete low-dimensional model independent of the mesh motion technique and the turbulence model used. At the *online phase*,

- POD and *radial basis functions* (RBF) are put together to reconstruct the mesh motion and also reduce the computational time due to the grid motion;
- POD and Neural Networks are combined to predict the Eddy viscosity.

## 1- Fluid and structure motions

Considering the incompressible Navier-Stokes in ALE formulation:

$$\nabla \cdot \bar{\mathbf{u}} = 0$$

$$\frac{\delta \bar{\mathbf{u}}}{\delta t} + \nabla \cdot (\bar{\mathbf{u}} \otimes (\bar{\mathbf{u}} - \mathbf{u}_g)) = \nabla \cdot \left[ \frac{1}{\rho} \nu_{eff} \nabla \bar{\mathbf{u}} \right] - \frac{1}{\rho} \left( \nabla \bar{p} + \frac{2}{3} \rho \nabla k \right).$$

In addition, **space conservation law** is enforced, plus *initial and boundary conditions*.

- $\frac{2}{3} \rho \nabla k$  is the normal stresses arising from the Boussinesq hypothesis;
- $\nu_t$  is the so-called turbulent viscosity and  $\nu_{eff} = \nu + \nu_t$  is the effective viscosity;
- $k$  is the turbulent kinetic energy;
- $\mathbf{u}_g$  is the grid velocity.

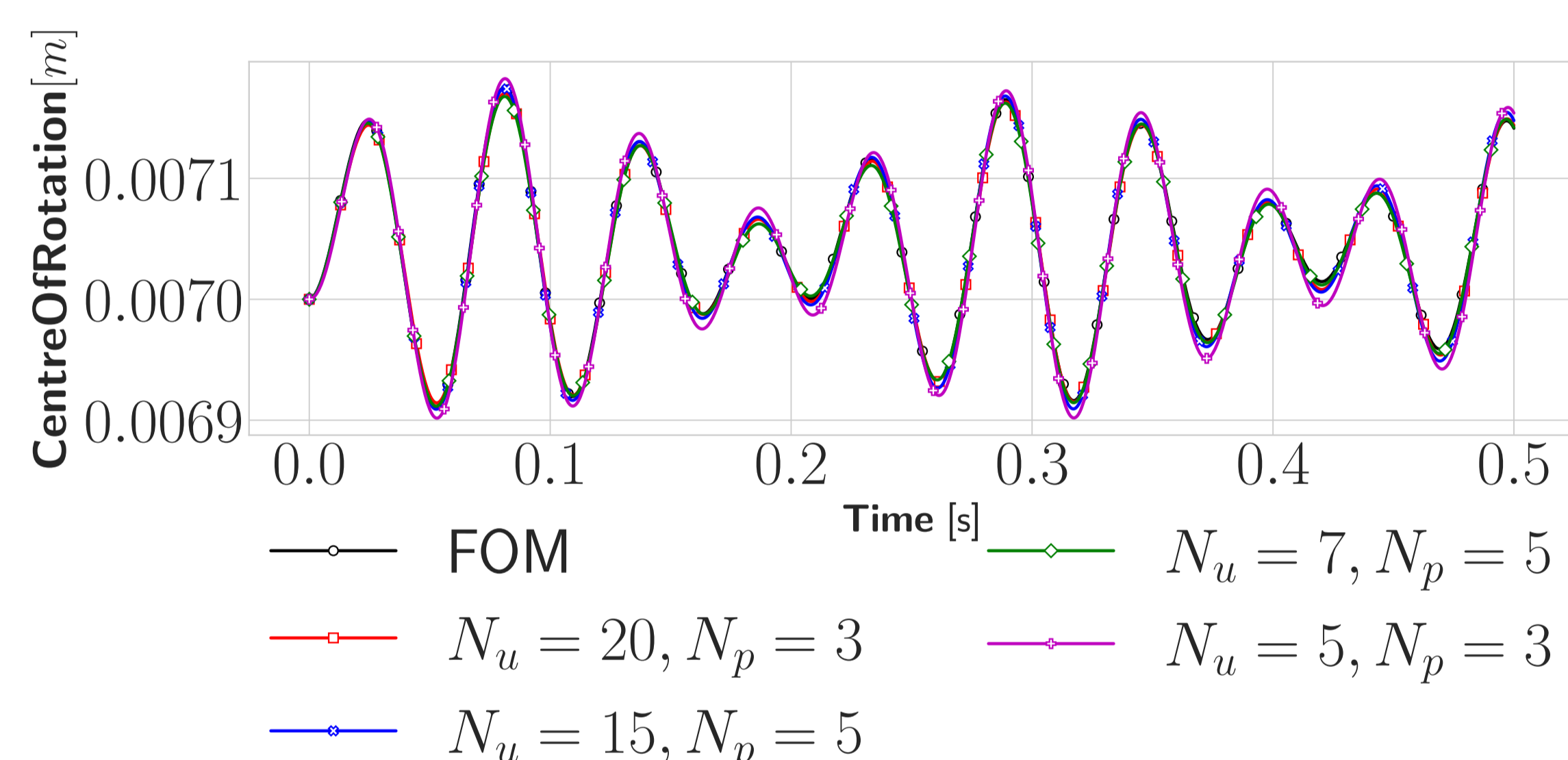
$$m \ddot{h} + c_h \dot{h} + k_h h - m b \ddot{\theta} \cos \theta + m b \dot{\theta}^2 \sin \theta = F_h(t)$$

$$\mathbf{I}_\theta \ddot{\theta} + c_\theta \dot{\theta} + k_\theta \theta - m b \dot{h} \cos \theta = \mathbf{M}_\theta(t)$$

$F_h$  is the **lift force**,  $\mathbf{I}_\theta$  is the moment of inertia of the foil,  $\theta$  is the **pitch** rotation,  $h(t)$  is the **plunge** displacement,  $b$  is the distance between the pivot location and the center of mass. The structural **stiffness** of the plunge and pitch is designated by  $k_h$  and  $k_\theta$ .

## 2A - Time series of the plunge

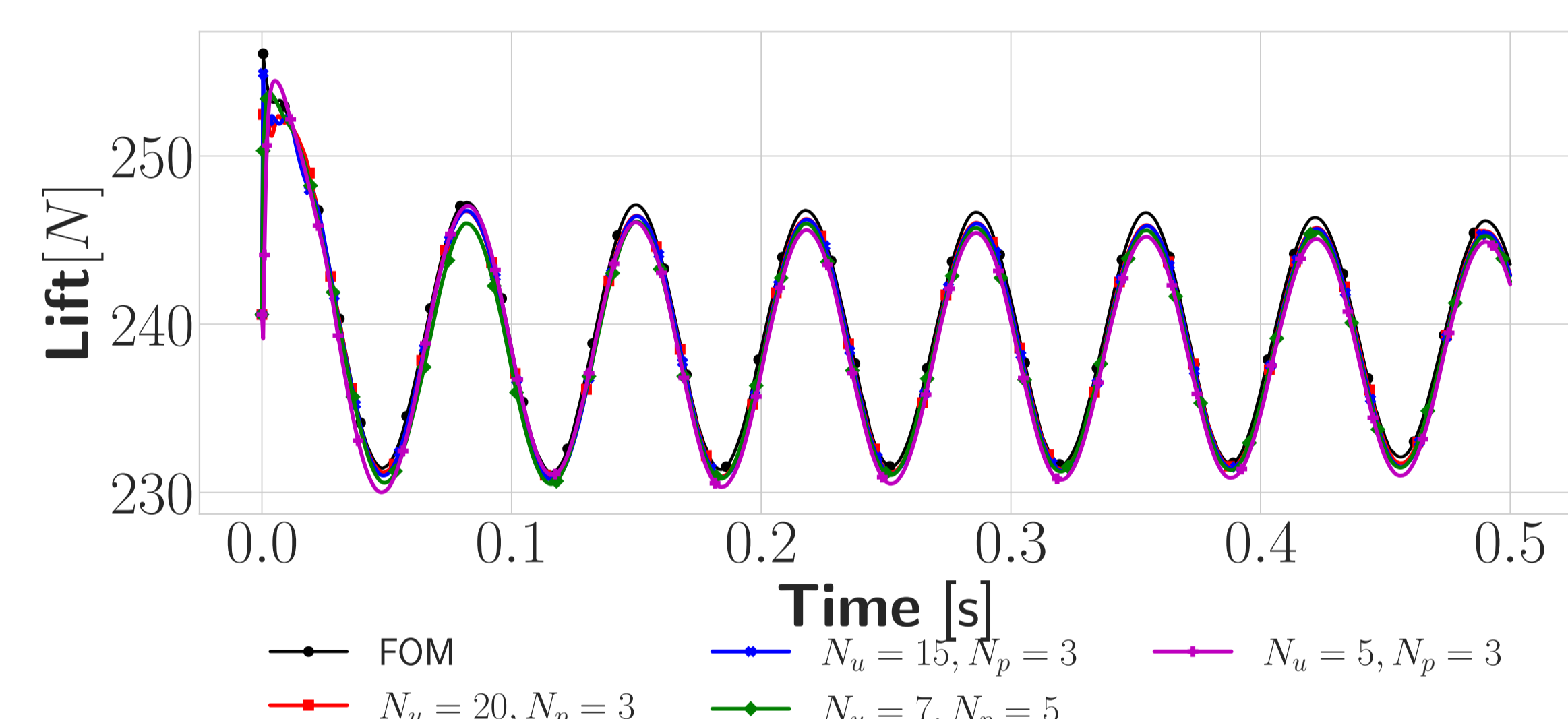
The time series of the displacement of the center of rotation.



**Figure 1:**  $N_u$  and  $N_p$  are the number of spatial modes used to reconstruct for both the velocity and pressure.

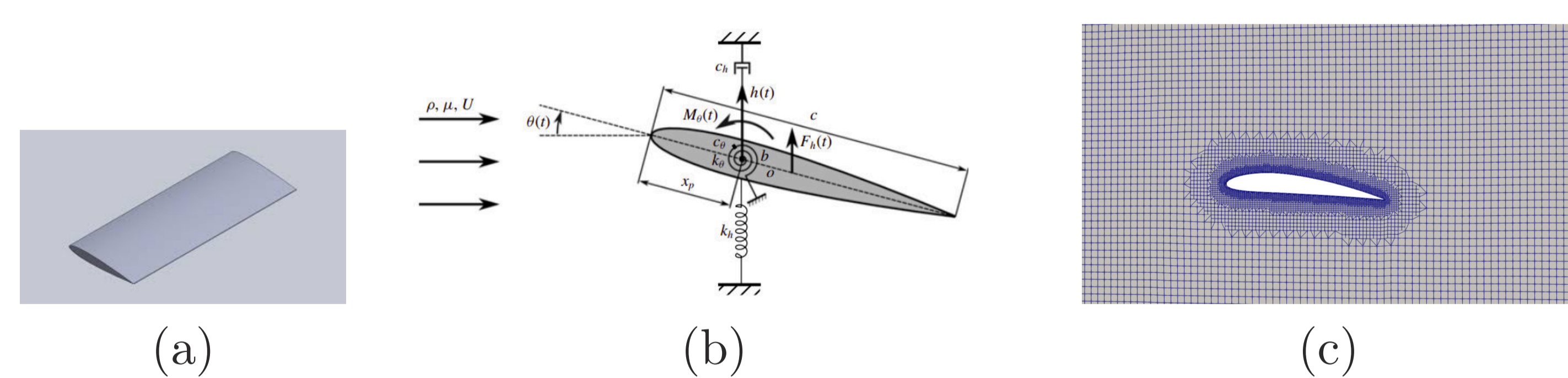
## 2B - Lift forces analysis

The figure shows the effect of the number of modes on the accuracy.



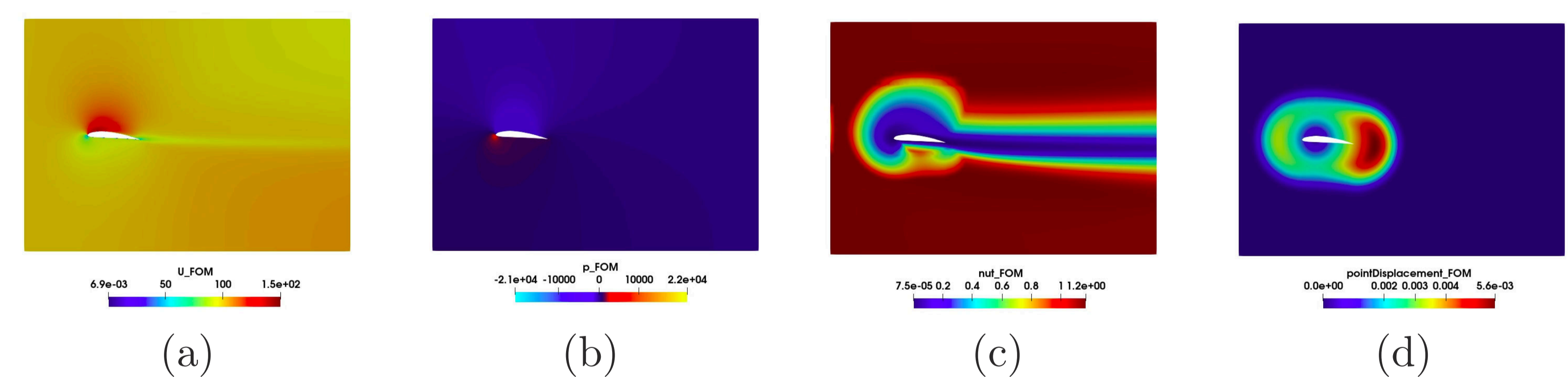
**Figure 3:**  $N_u$  and  $N_p$  are the number of spatial modes used to reconstruct for both the velocity and pressure

## 3A - Definition of the test case




**Figure 2:** Naca 0012 3D airfoil model (a), a schematic of the fluid-structure system considered (b): a foil allowed to undergo 2 degrees of freedom fully passive plunging and pitching motion, and zoom mesh around the airfoil (c)

## 3B - Quantities of interests




**Figure 4:** velocity (a), pressure (b), Eddy viscosity (c), and point cloud displacement of the grid (d)

## 4 - Computational science and engineering softwares: [mathlab.sissa.it/cse-software](http://mathlab.sissa.it/cse-software)



**PyGeM**  
github.com/mathLab/PyGeM  
mathlab.github.io/PyGeM

PyGeM is a python package using Free Form Deformation, Radial Basis Functions, and Inverse Distance Weighting to morph complex geometries.




**PyDMD**  
github.com/mathLab/PyDMD  
mathlab.github.io/PyDMD

PyDMD is a Python package that uses Dynamic Mode Decomposition for a data-driven model simplification based on spatiotemporal coherent structures.



**ITHACA-FV**  
github.com/mathLab/ITHACA-FV  
mathlab.github.io/ITHACA-FV

ITHACA-FV is an implementation in OpenFOAM of several reduced-order modeling techniques based on the Finite Volume Method.



**EZYRB**  
github.com/mathLab/EZYRB  
mathlab.github.io/EZYRB

EZYRB is a Python library for data-driven (non-intrusive) model order reduction with POD with interpolation.

## References and Acknowledgements

- [1] T. Lieu, C. Farhat, and M. Lesoinne. Reduced-order fluid/structure modeling of a complete aircraft configuration. *Computer methods in applied mechanics and engineering*, 195(41-43):5730–5742, 2006.
- [2] V. N. Ngan, G. Stabile, A. Mola, and G. Rozza. A reduced-order model for segregated fluid-structure interaction solvers based on an ALE approach. *arXiv preprint arXiv:2305.13613*, 2023.

This work was partially funded by European Union Funding for Research and Innovation — Horizon 2020 Program — in the framework of European Research Council Executive Agency: H2020 ERC CoG 2015 AROMA-CFD project 681447 "Advanced Reduced Order Methods with Applications in Computational Fluid Dynamics" P.I. Professor Gianluigi Rozza and by PRIN "Numerical Analysis for Full and Reduced Order Methods for Partial Differential Equations" (NA-FROM-PDEs) project.